**MEDA, AN ENVIRONMENTAL AND METEOROLOGICAL PACKAGE FOR THE MARS 2020 MISION.** J.A. Rodriguez-Manfredi<sup>1</sup>, M. de la Torre<sup>2</sup>, N.T. Bridges<sup>3</sup>, P. G. Conrad<sup>4</sup>, F. Ferri<sup>5</sup>, M. Genzer<sup>6</sup>, F. Gómez-Gómez<sup>1</sup>, J. Gómez-Elvira<sup>1</sup>, A-M. Harri<sup>6</sup>, O. Kemppinnen<sup>6</sup>, M. Lemmon<sup>7</sup>, G. Martínez<sup>8</sup>, S. Navarro<sup>1</sup>, C. Newman<sup>9</sup>, S. Pérez-Hoyos<sup>10</sup>, O. Prieto-Ballesteros<sup>1</sup>, M. Ramos<sup>11</sup>, A. Saiz-López<sup>12</sup>, A. Sánchez-Lavega<sup>10</sup>, J.T. Schofield<sup>2</sup>, E. Sebastian<sup>1</sup>, M. Smith<sup>4</sup>, L.K. Tamppari<sup>2</sup>, and the MEDA team. <sup>1</sup>Centro de Astrobiología (INTA-CSIC), Madrid, Spain; <sup>2</sup>Jet Propulsion Laboratory, Pasadena, CA 91109; <sup>3</sup>Applied Physics Laboratory, 11100 Johns Hopkins Road, Laurel, MD 20723; <sup>4</sup>NASA Goddard Space Flight Center, 8800 Greenbelt Rd, Greenbelt, MD 20771; <sup>5</sup>Università degli Studi di Padova, Centro Di Ateneo Di Studi E Attivita' Spaziali "Giuseppe Colombo" (CISAS), Padova Italy; <sup>6</sup>Finnish Meteorological Institute, Erik Palménin aukio 1, 00560 Helsinki, Finland; <sup>7</sup>Texas A&M University, College Station, TX 77843; <sup>8</sup>University of Michigan, 500 S State St, Ann Arbor, MI 48109; <sup>9</sup>Ashima Research, 600 S. Lake Ave, Suite 104, Pasadena, CA 91106; <sup>10</sup>University of Basque Country, Bilbao, Spain; <sup>11</sup>University of Alcalá, Plaza de San Diego, 28801, Alcalá de Henares, Spain; <sup>12</sup>Institute Physical-Chemistry Rocasolano, CSIC, Serrano 119, 28016, Madrid, Spain.

**Introduction:** The *Mars Environmental Dynamics Analyzer* (MEDA) has been selected as one of the seven payload investigations onboard the NASA Mars 2020 rover mission [1].

MEDA is a contributed legacy of the Mars Science Laboratory's REMS (Rover Environmental Monitoring Station), a modular sensor set designed to address two Mars 2020 investigation goals: "characterization of dust size and morphology", and "surface weather measurements", while exploiting the REMS [2] and PanCam/HazCam [3] heritage. MEDA is designed to be more than a dust characterization and MET station package; it offers synergies with the goals of other payload investigations on Mars 2020 (specially with the MOXIE investigation *–Mars Oxygen ISRU Experiment-*, designed to produce oxygen from Martian atmospheric carbon dioxide), as well as respond to more general Mars Program objectives, and Mars Strategic Knowledge Gap investigations [4].

MEDA will continue the long line of Mars meteorological packages from Viking, to Pathfinder, Phoenix, MSL, and, hopefully, InSight, thereby expanding surface environmental measurements to span nearly half a century. The MEDA team includes scientists and engineers from most of these missions, providing a wealth of experience that will contribute to successful development and operations.

**MEDA goals and objectives:** MEDA's goal is to help understand the Martian surface conditionsby sampling the near surface environment. Its sensors will enable comparisons to the measurements obtained at other locations previously explored on Mars. The MSL REMS heritage additionally permits easier comparisons to the meteorological station presently operating in Gale Crater. The measurement objectives are:

a. The physical and optical properties of the local atmospheric aerosols. Particle abundance, size distribution, shape, phase function, and how these optical properties relate to the meteorological cycles (diurnal, seasonal, interannual).

b. The conditions leading to dust lifting and how the aerosol diurnal cycle responds to the local atmospheric wind regimes.

c. To determine how the current environmental pressure, temperature, relative humidity, solar radiation, net infrared radiation, and winds at the landing site differ from those at the Viking, Phoenix, Pathfinder, and Curiosity locations.

d. Probe the relationship between the surface environment and the large-scale dynamics observed from orbiting instruments.

e. Investigate the energy and water fluxes between the surface and the lower atmosphere of Mars near the rover.

f. Measure the annual cycles of the solar UV, visible and NIR radiation on the surface of Mars.

g. Obtain the environmental context for weathering and preservation potential of a possible cache sample.

h. Investigate how pressure, humidity, temperature and winds influence the ISRU engineering efficiency?

i. Form linkages between the MEDA observations and models extrapolated to the Martian surface.

**MEDA's concept:** MEDA will carry a dust and optical radiation sensor, pressure sensor, relative humidity sensor, wind sensor, air temperature sensors, thermal infrared net flux and ground temperature sensor. They monitor autonomously around the clock. The solar radiation sensor is designed to track direct and diffuse radiation in a geometry that characterizes both, the prevailing environmental dust properties [5,6], and the behavior of solar radiation on subdiurnal time scales. This helps constraint and model the impact of solar radiation on local photochemistry, thus supporting assessments of the preservation potential for organics on a cache sample.

Table 1 shows the investigation traceability matrix listing and linking the MEDA sensors, to observables and goals.

Mars 2020 AO	MEDA Sci- ence Goals	MEDA Scientific Measurement Requirements		
Threshold Science Objectives		Observables	Physical Parameters	
Objective D.2 Characteri- zation of	Determine optical prop- erties of Martian dust and how it changes with time and meteorologi- cal conditions	Angular de- pendence of diffuse VIS solar light (sideways cone radiation sen- sor)	Dust phase func- tion, optical shape and size distribu- tion	
Dust size and mor- phology		Angular de- pendence of scattered light near Sun ( <u>CCD</u> )	Optical depth and size distribution	
	Determine the dynamics of the Martian	Thermistors resistance ( <u>thermocouple)</u>	Air Temperature	
	environment Thermal environment	Piezoelectric Capacitor ( <u>pressure</u> <u>sensor)</u>	Atmospheric Pressure	
	Global dust and CO2 cycles	Heat fluxes to maintain con- stant ∆T be- tween refer-	Wind velocity	
Objective D.3	Wind charac- teristics	ence tempera- ture and 4 dice (hot plate anemometers)	Wind direction	
Make Sur- face Weather Measure- ments to Validate Global Atmospheric Models	Surface temperature and fluxes with the low atmosphere	Thermal radia- tion in several bands ( <u>ground</u> <u>looking thermal</u> <u>IR sensors</u> )	Regolith surface temperature and emissivity.	
	Characterize the solar radiation environment at the surface of Mars and its potential effects on life	UV-VIS-NIR irradiance fluxes ( <u>whole</u> <u>sky FOV pho-</u> <u>todiodes with</u> <u>filters)</u>	UV flux and O3	
		Thermal radia- tion in several bands ( <u>upward</u> <u>TIR sensors</u> )	Air temperature at 0.6Pa	
	Understand the Martian hydrological cycle	Capacitance ( <u>relative humid-</u> ity sensor)	Relative humidity	

Table 1. MEDA baseline	design	flow	from	goals	and
objectives.					

Resolving dust and environmental variables over many time scales is required to understand (a) the predictive capabilities of models of the near surface environment on Mars, and (b) assess how the environment affects operational ISRU and rover engineering cycles. Therefore, MEDA's operations concept is to work autonomously and continuously with programmable continuuous temporal coverage and a variable sampling rate, even during rover sleep periods.

It is intended to accommodate MEDA's wind, humidity, air temperature and net thermal infrared radiation flux sensors on a dedicated and deployable mast placed in the port-front corner of the rover deck, to protect them from the rover geometry and thermal influences. Dust and solar radiation sensors will be placed over the rover deck, to maximize their unobstructed fields of view.

## **References:**

[1] NASA Mars 2020 Press Release and Press Kits, http://www.nasa.gov/press/2014/july/nasa-announcesmars-2020-rover-payload-to-explore-the-red-planet-asnever-before/

[2] Gómez-Elvira, J. et al. (2012), /SSR, 170,/ 583-640.

[3] Bell III, J. F., S. W. Squyres, K. E. Herkenhoff, et al., The Mars Exploration Rover Athena Panoramic Camera (Pancam) investigation, J. Geophys. Res. 108(E12), doi:10.1029/2003JE002070, 2003.

[4] P-SAG (2012) Analysis of Strategic Knowledge Gaps Associated with Potential Human Missions to the Martian System: Final report of the Precursor Strategy Analysis Group (P-SAG), D.W. Beaty and M.H. Carr (co-chairs) + 25 co-authors, sponsored by MEPAG/SBAG, 72 pp., posted July 2012, by the Mars Exploration Program Analysis Group (MEPAG) at http://mepag.jpl.nasa.gov/reports/

[5] Dubovik, O., and M. D. King (2000), A flexible inversion algorithm for retrieval of aerosol optical properties from Sun and sky radiance measurements, J. Geophys. Res., 105(D16), 20673–20696, doi:10.1029/2000JD900282 <a href="http://dx.doi.org/10.1029/2000JD900282">http://dx.doi.org/10.1029/2000JD900282</a>

[6] Lemmon, M. T., M. J. Wolff, M. D. Smith, R. T. Clancy, D. Banfield, G. A. Landis, A. Ghosh, P. H. Smith, N. Spanovich, B. Whitney, P. Whelley, R. Greeley, S. Thompson, J. F. Bell III, and S. W. Squyres. 2004. Atmospheric Imaging Results from the Mars Exploration Rovers: Spirit and Opportunity. /Science/ 306