## Overview

On this program, Creare will develop and demonstrate an innovative cryocooler that produces refrigeration at temperatures of 30 to 70 K and rejects heat at a temperature of 150 to 210 K with extremely high efficiency. The heat rejected can be absorbed by an upper stage cryocooler or rejected to space through a small cryo-radiator.
The demonstration system will include a combination of new and existing components. The Phase II testing will be structured to achieve a TRL of at least 5, and will include cryogenic performance testing and launch vibration testing. The cryocooler would be space-qualified during a follow-on Phase III project.

## System Specifications

- Net refrigeration: 300 mW at 35K
- CCE Input power: 20 W at 28 VDC
- Cryo-compressor AC Input power 8.9W

Efficiency: 11\% of Carnot

- Total mass: 6.2 kg
- TMU: 2.8 kg
- CCE: 1.4 kg
- Cryo-Radiator: 1.0 kg
- Misc: 1.0 kg
- Radiation tolerance
- 300 kRad at parts level
- 0.125 in . aluminum shel
- Radiator area: $0.8 \mathrm{~m}^{2}$ necessarily reflect the views of NASA.


## Conceptual System Layout

Cryocooler Control Electronics (CCE)


Single Stage Reverse Brayton Thermodynamic Cycle


## Demonstration System



## Potential NASA Applications

Future satellites, probes and astronomical observatories utilizing superconducting bolometers, and infrared, far infrared, submillimeter and X -ray detectors.

