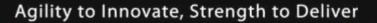
Flash Lidar for Planetary Missions

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Planetary Applications for Flash Lidar



- Topographic mapping from orbit about a planetary body or in a following orbit with a small body
- 3-D surface information available real-time under any lighting condition and while in motion

AUTONOMOUS RENDEZVOUS AND DOCKING

 Docking with a carrier spacecraft or rendezvous with an asteroid or other non-cooperative target

ENTRY, DESCENT AND LANDING

- Initial Entry: LIDAR used for altimetry and velocimetry. With addition of an imager, Terrain Relative Navigation may be performed, even under low light conditions.
- Terminal Decent: LIDAR beam diverted to illuminate entire field of regard. Allows addition of Hazard Detection and Avoidance for landing.

High resolution

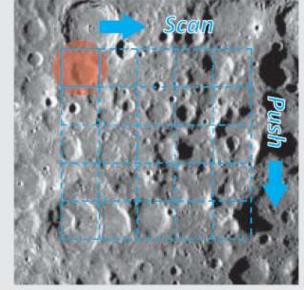
maps of terrain and rock surfaces



Scanning vs. Flash Lidar

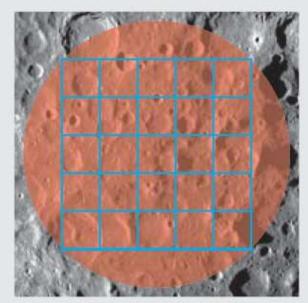
- Range to target is determined by measuring the time of flight of a laser pulse from source to target and back to detector
- Intensity of returned light is also measured
- Difference between scanning and flash lidar lies in illumination and detection approach

(a) Scanning Pushbroom Lidar



- Spot size matches single pixel or detector field of view
- Spot scanned to make multiple
 Entire scene is mapped with measurements at adjacent locations to build up a swath
- Process is repeated for the next swath

(b) Flash Lidar



- Spot size matches 2-D array detector field of view
- one laser pulse



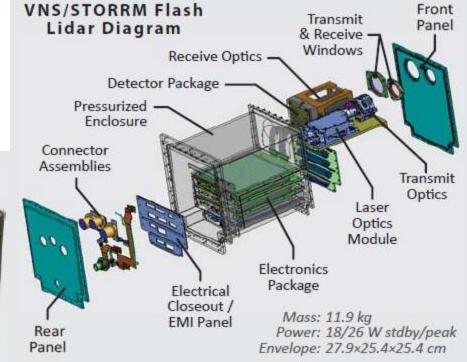
- Fast Frame Rate \rightarrow Insensitivity to target motion, stray light and glint
 - Entire scene imaged at once no scanning required
 - Entire array of pixels (frame) is correlated in space and time with each laser pulse
 - Each pixel is read out in parallel via special Read Out Integrated Circuit (ROIC)
- Large Dynamic Range \rightarrow High resolution imagery of rough surfaces
 - Each pixel is triggered and read independently, allowing a large dynamic range
- High Reliability \rightarrow Low risk, long life
 - No moving parts
 - Long life lasers rated for over 2 billion shots
 - Large array offers detector redundancies
- Lower Mass and Volume
 - No scanning mechanisms
- Simplified Integration and Test
 - No precision alignment required



Orion Vision Navigation System (VNS) Flash Lidar

- Developed as a docking sensor for the Orion/MPCV capsule with the ISS
- Full frame flash lidar, max 30 Hz frame rate
- CMOS InGaAs PIN detector array (256x256)
- Optically switchable diffusers to create 12 or 20 deg beam divergence
- 2.5 mJ Nd:YAG passively Qswitched pulsed laser
- Optical parametric oscillator to create eye-safe 1572 nm output
- 33 mm receive aperture; 20 deg FOV

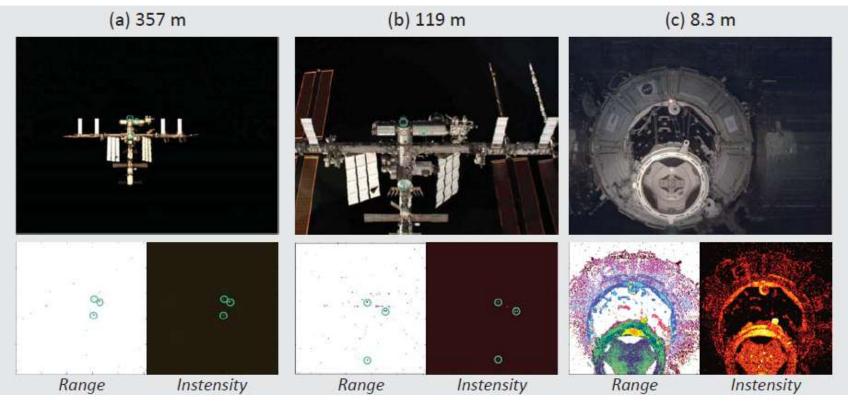






Sensor Test for Orion Relative-navigation Risk Mitigation (STORRM)

- VNS flew on Space Shuttle mission STS-134 in May 2011
- VNS mounted in the cargo bay was activated as the shuttle approached the ISS
- VNS successfully acquired the ISS at a range of 5.6 km, exceeding 5.0 km requirement
- The lidar never lost lock the entire time the ISS target was in the field of view
- Provided real-time range and bearing information, proved the conceptual approach and elevated the hardware to TRL 7

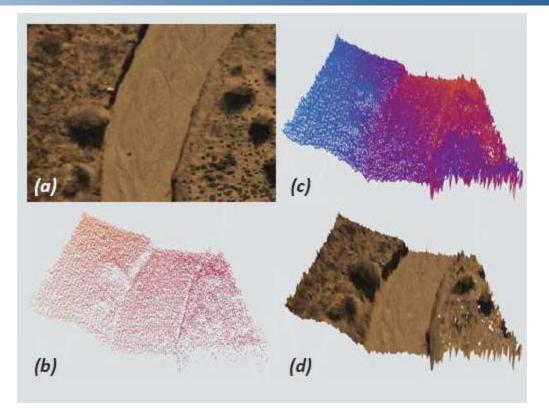




- ISS docking targets are ~95% reflective, while planetary targets have ~3-20% albedo
- More signal is needed to use VNS/STORRM for planetary missions
- Low cost, low risk modifications
 - Remove OPO 1064 nm laser output with 60% more light output
 - Change diffusion filters on laser output → more laser energy per area (pixel)
 - Increase receive optics aperture → more light received per pixel but more mass
- Medium Cost & Risk modification
 - Increase laser power \rightarrow more light on target but more power and mass

Advanced Real-Time Processing

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- To provide true utility for advanced applications such as hazard avoidance, lidar data must be processed in real-time
- Flash has an advantage in that data is acquired quickly and the format is well suited for fast FPGA processing
 - (b) xyz-positions for each pixel (point cloud) are generated at 30 Hz
 - (c) a surface grid is generated for the frame
- The resulting DEM can be fed to GNC algorithms for landing or hazard avoidance



- (d) surface grid may be draped with a visible image from a camera or database for terrain relative navigation
- Entire process takes only milliseconds
- TRL 5 (system demonstrated on aircraft)





- Mature Flash Lidar technologies are available to provide important data for planetary missions
- Flash Lidar is particularly well suited for applications requiring fast data acquisition and read-out such as: descent and landing, hazard avoidance, rendezvous, encounters and docking, terrain relative navigation
- Ball is continuing to mature techniques and hardware for increased functionality

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- Radiometry
 - Target is actively illuminated with laser light and a narrow band filter on the detector \rightarrow insensitivity to ambient and stray light
 - Integration time for detector is fixed by laser pulse width
 - Radiometry determined by: (1) laser power (2) beam divergence (3) target albedo (4) aperture of receive optics (5) detector sensitivity
- Ranging
 - Each pixel has a minimum threshold detection (# photons) to register a signal
 - As long as the light incident on the detector is above the pixel threshold and below the pixel saturation, a range will be calculated → range accuracy is NOT a function of brightness
 - Thresholds are determined by adjusting detector gain
 - Different detector gains may be determined during calibration and commanded in flight operation
- Ranging Accuracy
 - Contributing factors include: ground calibration accuracy, repeatability, laser jitter, laser rise time jitter, digitization, mounting accuracy and thermal and structural distortions
 - Errors are dominated by calibration accuracy and [pixel to pixel] repeatability